

## ME 1020 Engineering Programming with MATLAB

Problem 9.10:

10.\* A rocket's mass decreases as it burns fuel. The equation of motion for a rocket in vertical flight can be obtained from Newton's law, and it is

$$m(t) \frac{dv}{dt} = T - m(t)g$$

where  $T$  is the rocket's thrust and its mass as a function of time is given by  $m(t) = m_0(1 - rt/b)$ . The rocket's initial mass is  $m_0$ , the burn time is  $b$ , and  $r$  is the fraction of the total mass accounted for by the fuel.

Use the values  $T = 48,000$  N,  $m_0 = 2200$  kg,  $r = 0.8$ ,  $g = 9.81$  m/s<sup>2</sup>, and  $b = 40$  s. Determine the rocket's velocity at burnout.

Problem setup:

$$F = ma$$

$$T - mg = ma$$

$$a(t) = \frac{T}{m(t)} - g$$

$$a(t) = \frac{T}{\left[m_0 \left(1 - \frac{rt}{b}\right)\right]} - g$$

Integrate the acceleration to find the velocity.

$$v(t) = \int_0^t a(t) dt + v(0) = \int_0^t \left\{ \frac{T}{\left[m_0 \left(1 - \frac{rt}{b}\right)\right]} - g \right\} dt + v(0)$$

The initial velocity for this problem is  $v(0) = 0$ . Use the Substitution Method to integrate the acceleration.

$$\text{Let } w = m_0 \left(1 - \frac{rt}{b}\right), \quad dw = -\left(\frac{m_0 r}{b}\right) dt, \quad dt = -\left(\frac{b}{m_0 r}\right) dw$$

$$\text{Limits: @ } t = 0, \quad w = m_0; \quad @ t = b, \quad w = m_0(1 - r)$$

$$v = \int_{m_0}^{m_0(1-r)} \left(\frac{T}{w} - g\right) \left[-\left(\frac{b}{m_0 r}\right) dw\right]$$

$$v = -\left(\frac{b}{m_0 r}\right) \int_{m_0}^{m_0(1-r)} \left(\frac{T}{w} - g\right) dw$$

$$v = -\left(\frac{b}{m_0 r}\right) (T \ln w - gw) \Big|_{m_0}^{m_0(1-r)}$$

$$v = -\left(\frac{b}{m_0 r}\right) \{ [T \ln(m_0(1-r)) - g(m_0(1-r))] - [T \ln(m_0) - g(m_0)] \}$$

$$v = -\left(\frac{b}{m_0 r}\right) \left\{ T \ln \left[ \frac{m_0(1-r)}{m_0} \right] - g[m_0(1-r) - m_0] \right\}$$

$$v = -\left(\frac{b}{m_0 r}\right) [T \ln(1-r) + g m_0 r]$$

Evaluate the velocity at burnout:

$$v = -\left(\frac{(40)}{(2200)(0.8)}\right) [(48,000) \ln(1-0.8) + (9.81)(2200)(0.8)] = 1363.3 \text{ m/s}$$

```
% Problem 9.10
clear
clc
disp('Problem 9.10: Scott Thomas')

T = 48000; %N
m0 = 2200; %kg
r = 0.8;
g = 9.81; %m/s^2
b = 40; %sec
% Create an anonymous function for a(t) = T/(m0*(1 - (r*t/b))) - g
aoft = @(t) (T./(m0*(1 - (r*t/b)))) - g;
N = 1001;
t = linspace(0,b,N);
a = aoft(t);
v(1) = 0.0;
for k = 1:N-1
v(k+1) = v(k) + 0.5*(t(k+1) - t(k))*(a(k) + a(k+1));
end
disp('Acceleration at burnout = ')
a(N)
disp('Velocity at burnout = ')
v(N)

plot(t,a, t,v), xlabel('t (sec)')
ylabel('Acceleration and Velocity')
title('Problem 9.10: Scott Thomas')
legend('a (m/s^2)', 'v (m/s)', 'Location', 'Northwest')
```

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Acceleration at burnout =

ans =  
9.9281e+01

Velocity at burnout =

ans =  
1.3634e+03

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